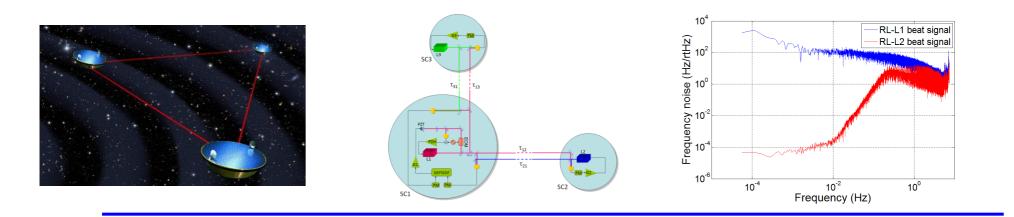


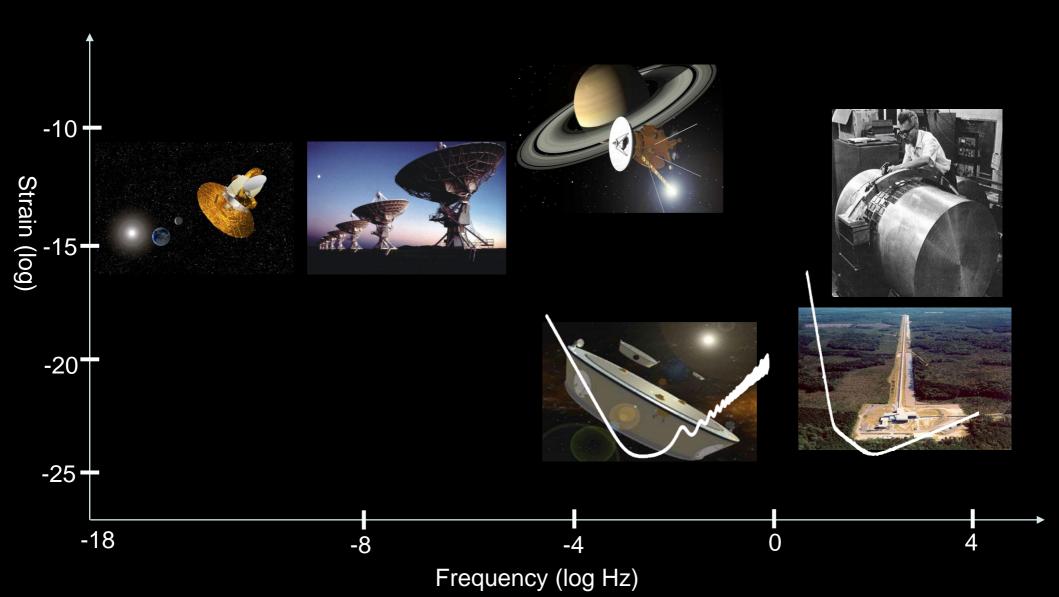
# Arm Locking for Laser Interferometer Space Antenna

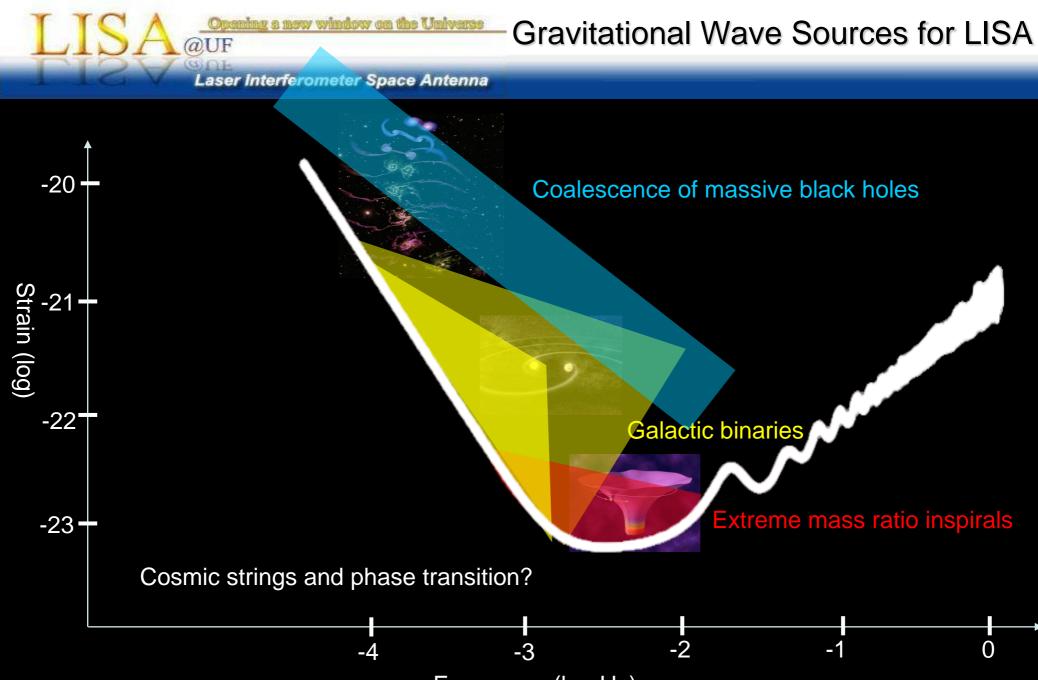


Yinan Yu University of Florida LISA Opening a new window on the Universe OUF

# **Gravitational Wave Astronomy**

Laser Interferometer Space Antenna





Frequency (log Hz)

# LISA Concept

#### Laser Interferometer Space Antenna

- Detect and observe GWs from 0.03 mHz to 1 Hz
- Stress Stress
- $^{\circ}$  Large separation 5 imes10 $^{9}$  m, variation  $\pm$ 1%

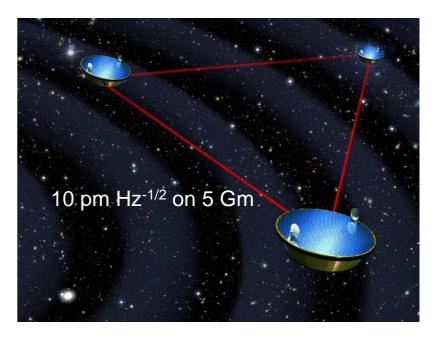
Sun

Trailing Earth by 20 degrees, out of the ecliptic by 60 degrees

Earth

5 Gm

1 AU



Key Technologies

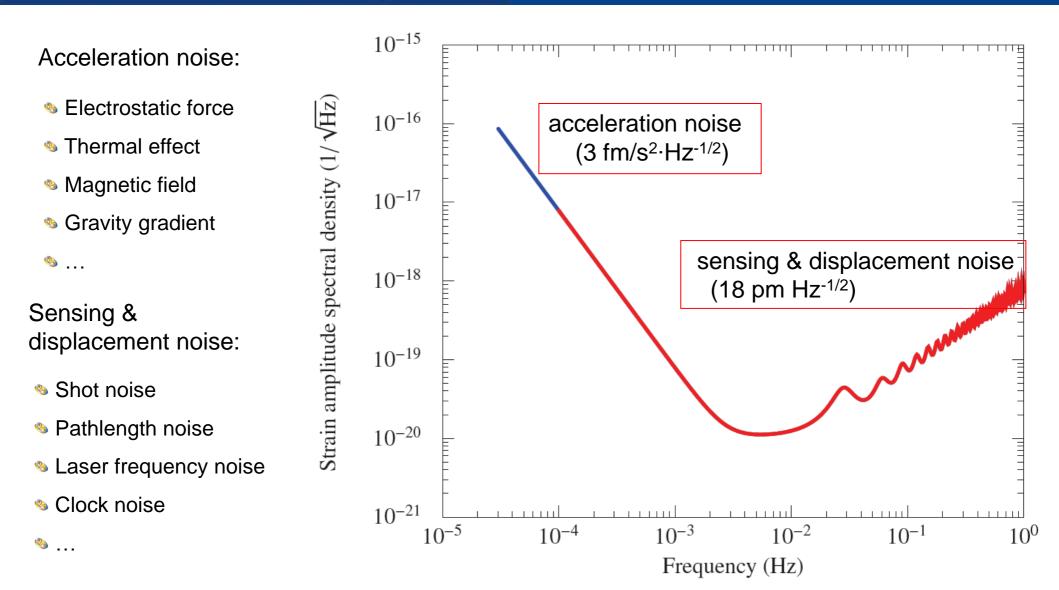
spicometer precision interferometry in space

In the second second

### Strain Sensitivity

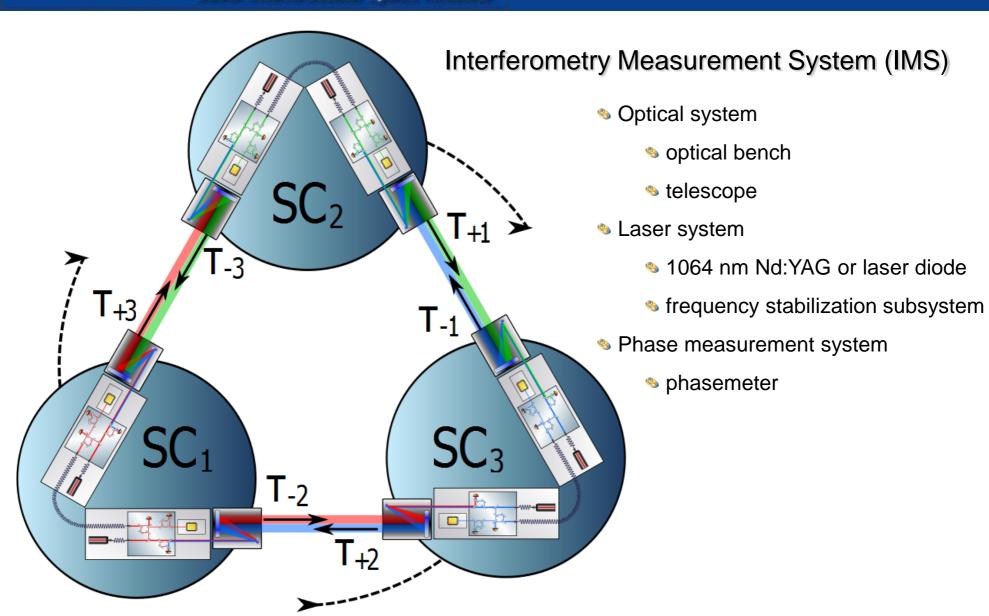
Laser Interferometer Space Antenna

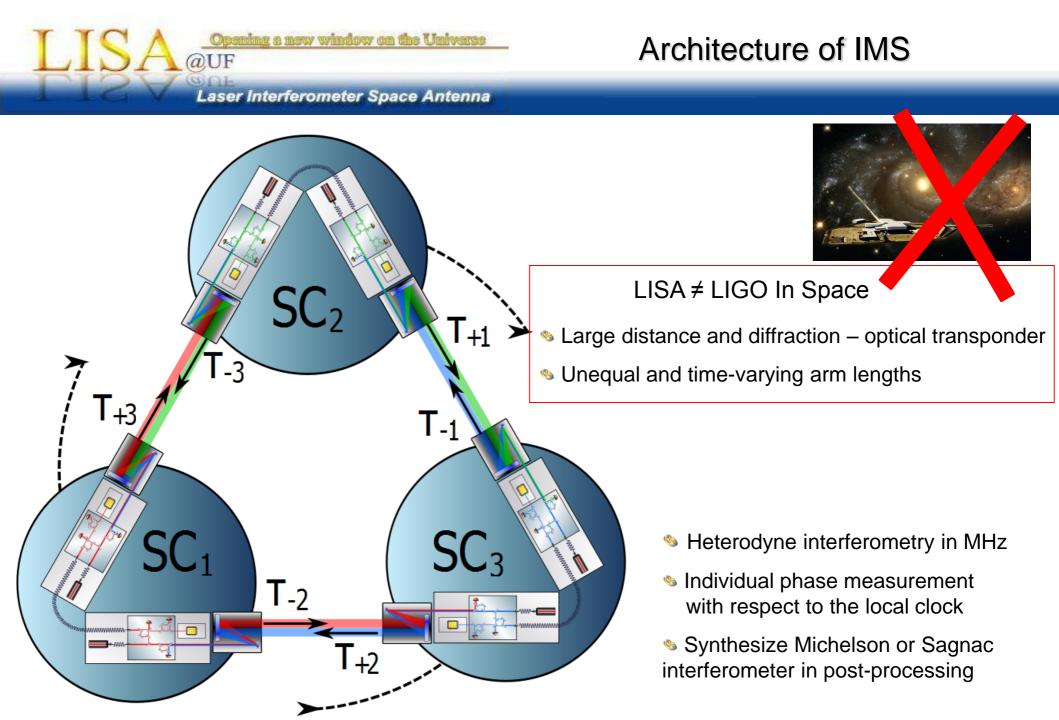
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# Architecture of IMS

Laser Interferometer Space Antenna





Laser Interferometer Space Antenna

# From Proof Mass to Proof Mass

PD<sub>R</sub>

short arm short arm long arm  $PD_R$ PD<sub>s</sub> PD<sub>s</sub>  $PD_{I}$  $PD_{I}$ 

Proof mass relative to the optical bench:

Reference interferometry (PD<sub>R</sub>)

Optical bench to optical bench

**OUF** 

 $\varphi_{\mathsf{R}}(t) = \Phi_{\mathsf{I}}(t) - \Phi_{\mathsf{a}}(t) + N_{\mathsf{a}}(t)$ Short arm interferometry (PD<sub>s</sub>)  $\varphi_{\rm S}(t) = \Phi_{\rm I}(t) - \Phi_{\rm a}(t) + \frac{2\pi}{\lambda}\Delta L_{\rm pm}(t) + N_{\rm a}(t)$ 

Long arm interferometry (PD<sub>I</sub>)  $\varphi_{L}(t) = \Phi_{I}(t) - \Phi_{f}(t - \tau_{ji}(t)) + N_{Trans}(t - \tau_{ji}(t)) + h_{ji}(t)$ 

Laser Interferometer Space Antenna

 $^{\star}$  Noise allocation for laser frequency noise:  $2 \ pm \ Hz^{-1/2}$ 

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Converted into phase noise:

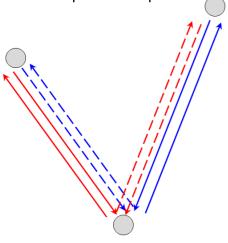
Converted into frequency noise:

$$\delta \varphi = \frac{2\pi}{\lambda} \delta L = 1.2 \times 10^{-5} \text{ rads } \text{Hz}^{-1/2}$$
$$\delta \nu = \frac{\delta L}{\Delta L} \nu = 1.1 \times 10^{-6} \text{ Hz } \text{Hz}^{-1/2}$$

Laser Frequency Control

- $\sim$  Free-running laser frequency noise: (10 kHz/f) Hz Hz<sup>-1/2</sup>
- Mow can we suppress the laser frequency noise by more than 12 orders of magnitude?

SC2&3: optical transponder



SC1: "beamsplitter'

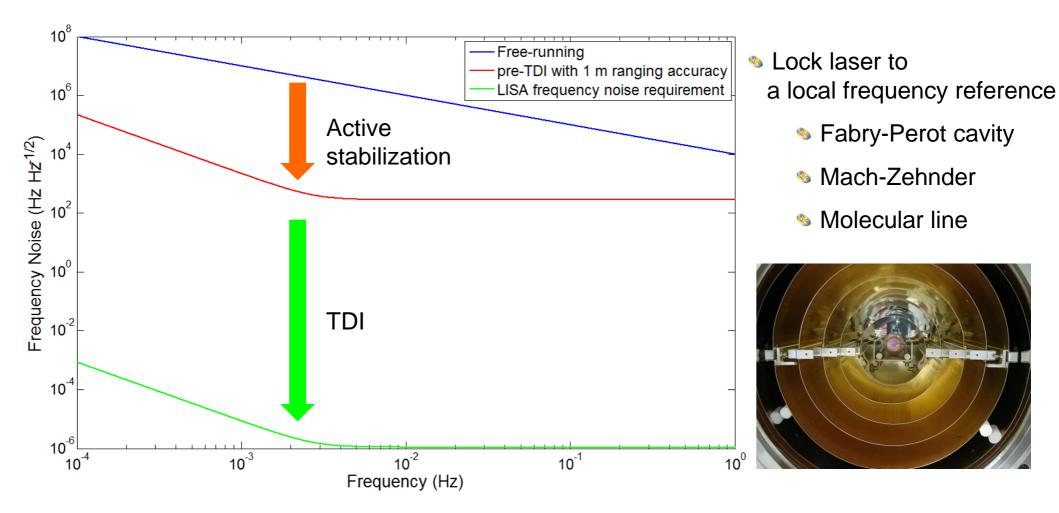
- Time Delay Interferometry
  Equal-arm Michelson interferometer in post-processing
- Require arm length knowledge Limited by 1 m ranging accuracy (if PRN ranging is used)

$$\delta \nu_{\rm pre-TDI}(f) < 282 \times \sqrt{1 + \left(\frac{2.8 \text{ mHz}}{f}\right)^4} \text{ Hz Hz}^{-1/2}$$

# **Active Laser Frequency Stabilization**

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### Arm Length as a Better Reference

#### Laser Interferometer Space Antenna

DIF

L3

 $\tau_{31}$ 

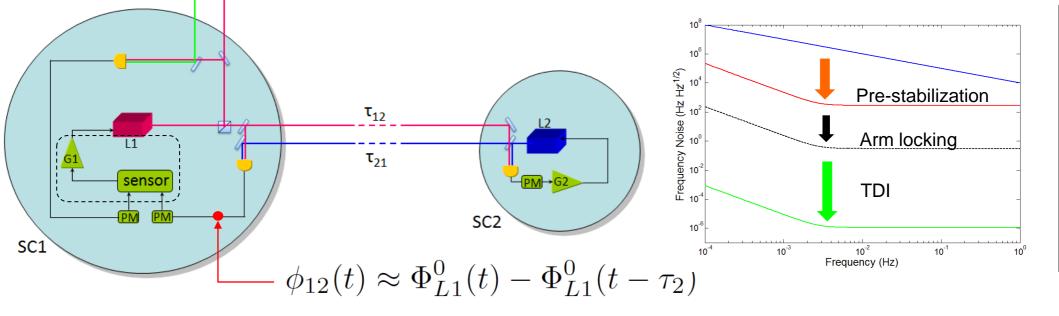
τ<sub>13</sub>

SC3

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Arm Locking: lock laser frequency to the LISA arms

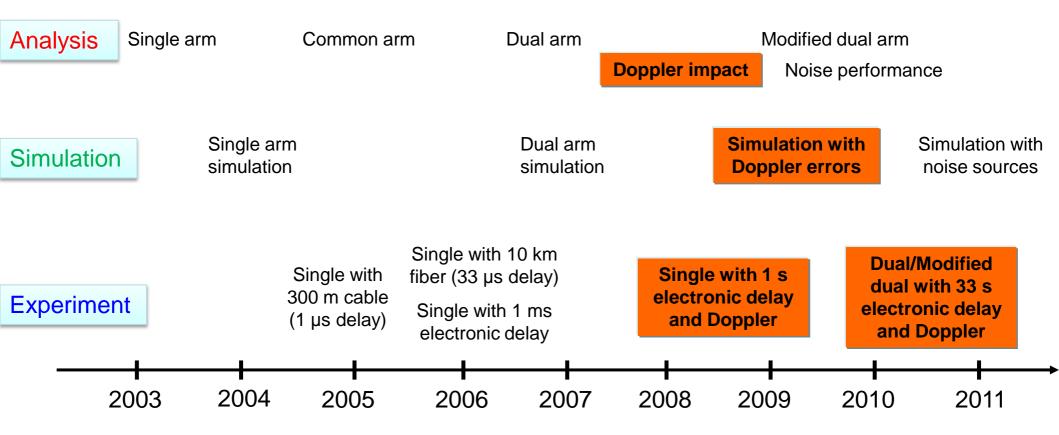
- $\sim$  More stable in the LISA frequency band:  $\delta L/L \sim 10^{-21} \text{ Hz}^{-1/2}$
- Stimate frequency noise from inter-SC phase measurements
- Scan be fully implemented in on-board data processing
- Scan be integrated with pre-stabilization system (requires tunable reference)



### **Arm Locking Chronicle**

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How to reproduce the 5 Gm arm length in lab?

**Doppler shifts** 

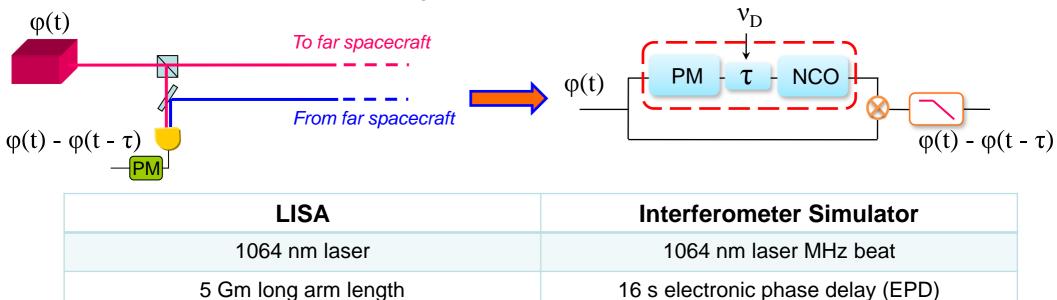
Constellation phase locking

Heterodyne interference

Phasemeter

Arm locking sensing & control law

Integration with pre-stabilized lasers



|                             | ) |
|-----------------------------|---|
| MHz frequency shifts in EPD |   |

Analog PLL

Analog mixing

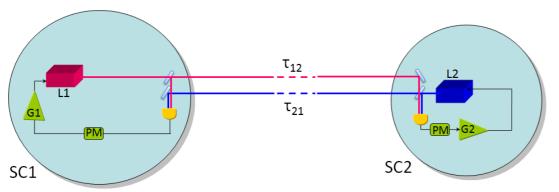
DPLL phasemeter prototype

Sensor & controller on FPGA

Heterodyne PLL / PZT cavity

# Single Arm Locking

Laser Interferometer Space Antenna

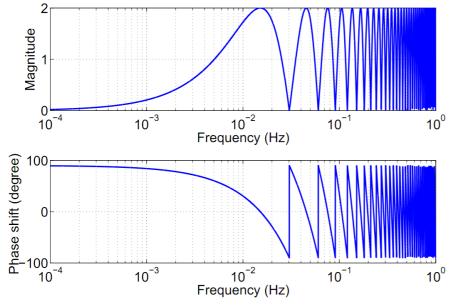


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- Mach-Zehnder interferometer with 10 Gm arm length mismatch
- Interferometer response:

$$P_{1i}(s) = rac{\phi_{1i}(s)}{\Phi^0_{L1}(s)} \approx 1 - e^{-s\tau_i}, \qquad i = 2,3$$

- insensitive at interferometer nulls
- Solution should be shou
- scontroller design needs to carefully keep enough phase margin
- control system is verified in simulations and experiments (with very short delay times)

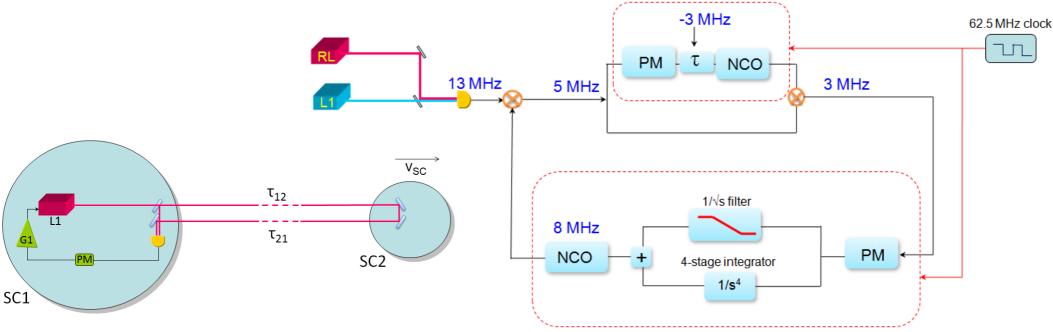


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### **Control System - Single**

DSP board 1 (EPD)

DSP board 2 (controller)



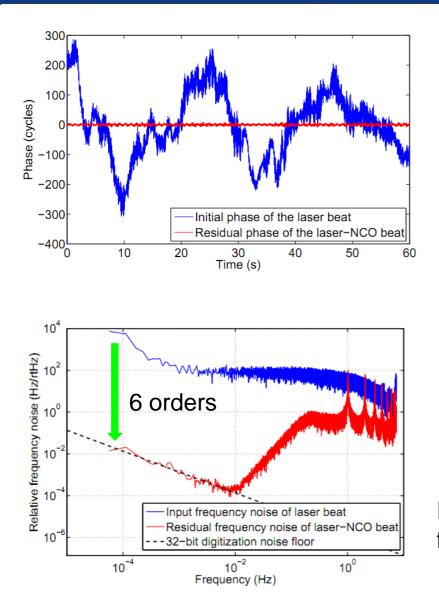
"Reflective mirror" at far SC

- NCO tracks the input phase noise
- s delay, Doppler shift
- scommon clock for both DSP boards

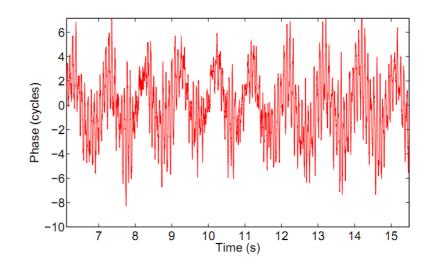
### Control System - Single

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- Sirst peak at 1 Hz
- 6 orders of magnitude suppression
- Limited by 32-bit digitization noise

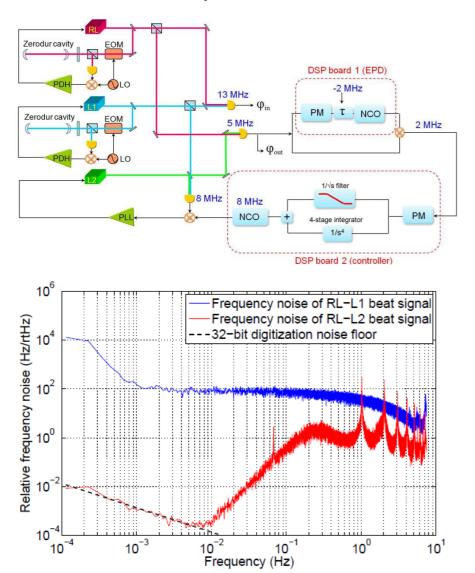
In realistic case, a 33 s delay would generate the first peak at 30 mHz, well inside the LISA band.

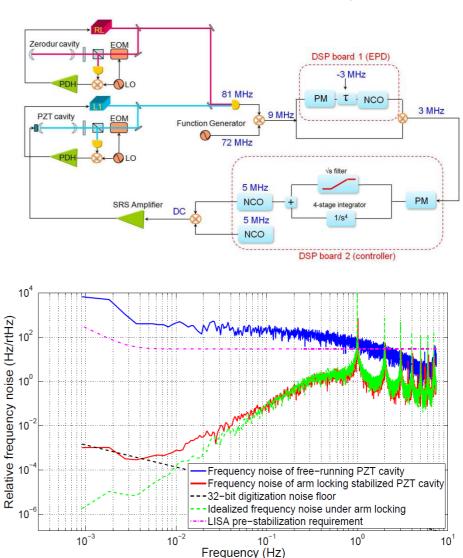
### Single AL with Pre-stabilized Laser

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Heterodyne PLL

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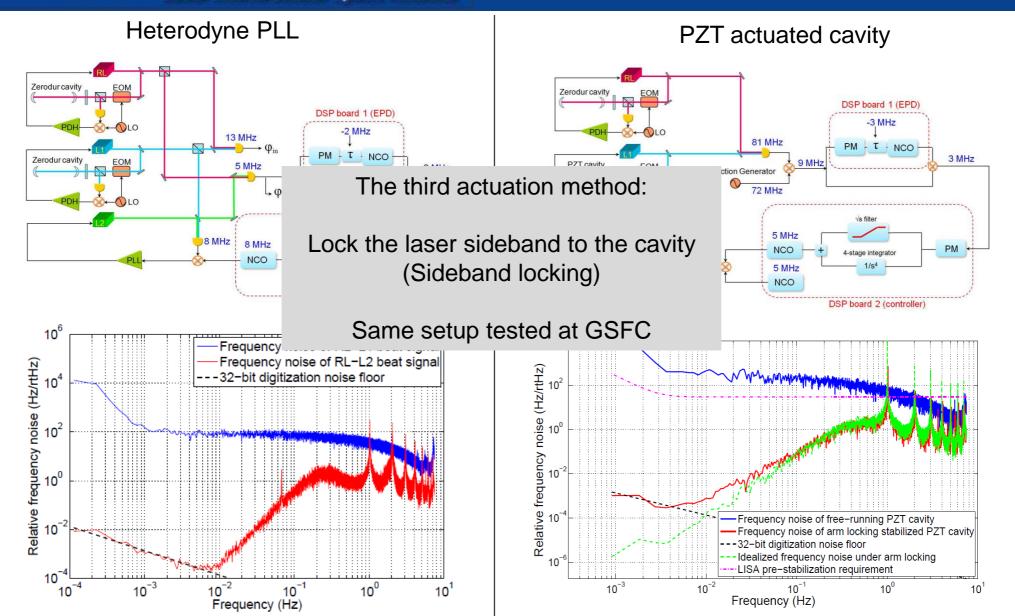


PZT actuated cavity

### Single AL with Pre-stabilized Laser

Laser Interferometer Space Antenna

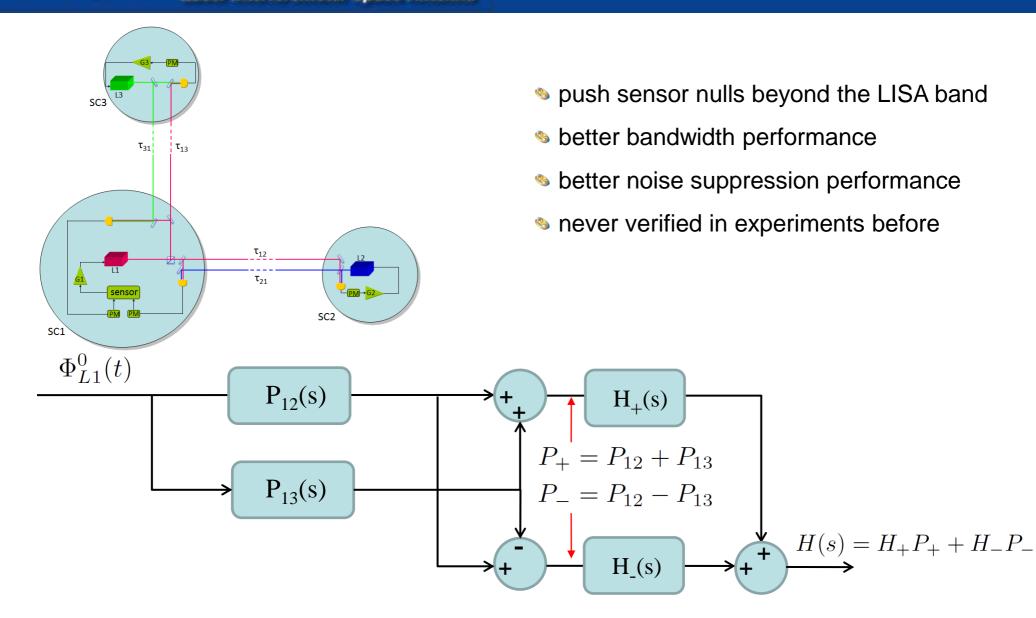
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#### Laser Interferometer Space Antenna

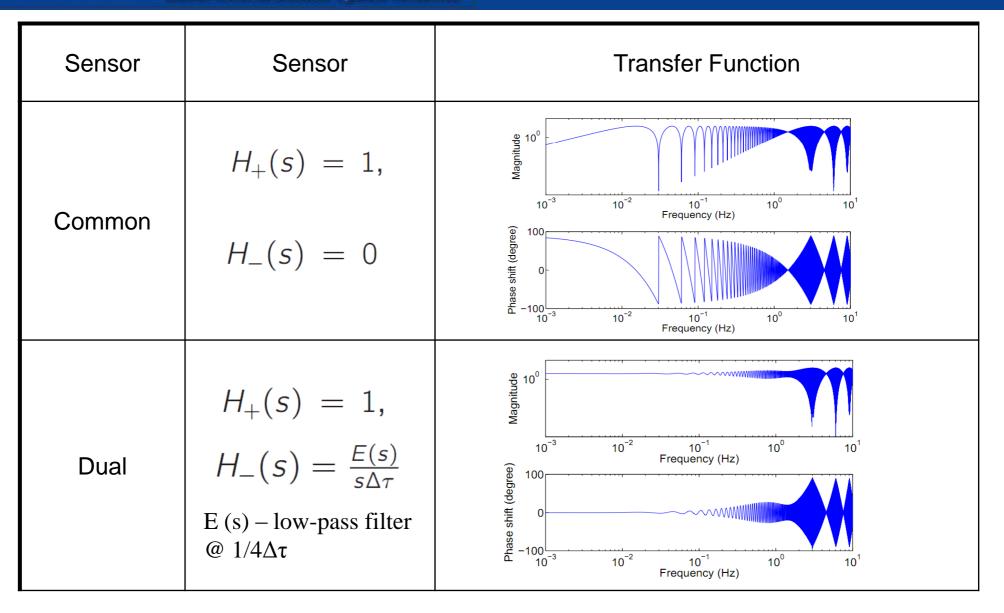


**Dual Arm Sensor** 

### **Common/Dual Arm Locking**

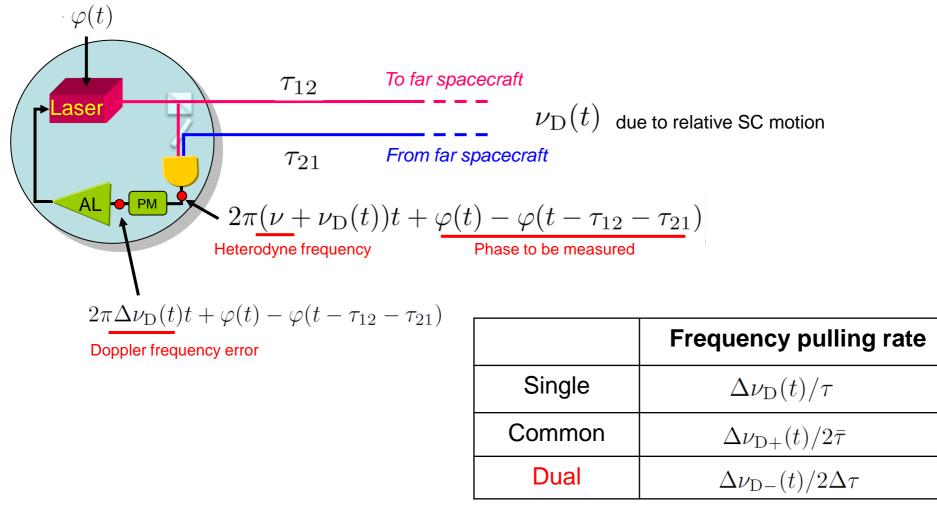
#### Laser Interferometer Space Antenna

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- <sup>∞</sup> Doppler shifts from relative SC motions ( $\pm$ 18 m/s  $\rightarrow$   $\pm$ 17 MHz)
- scause a Doppler error in phase measurements

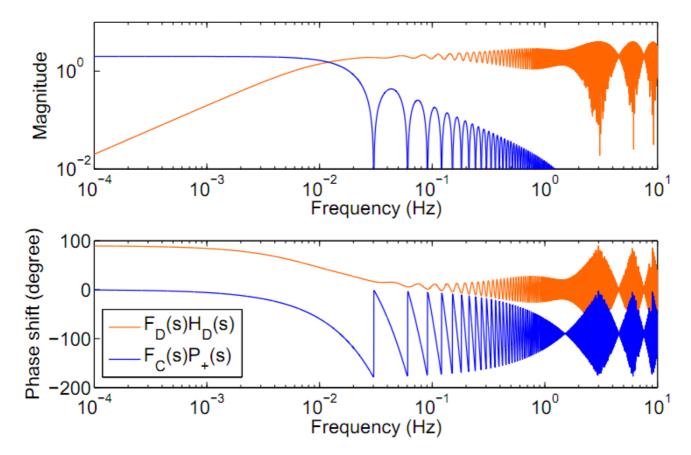




A combination of

 $\sim$  common arm sensor at f < 30 mHz (frequency pulling  $\Delta \nu_{\rm D+}(t)/2\bar{\tau}$  & noise advantage)

Solution of the sensor at f > 30 mHz (gain advantage)

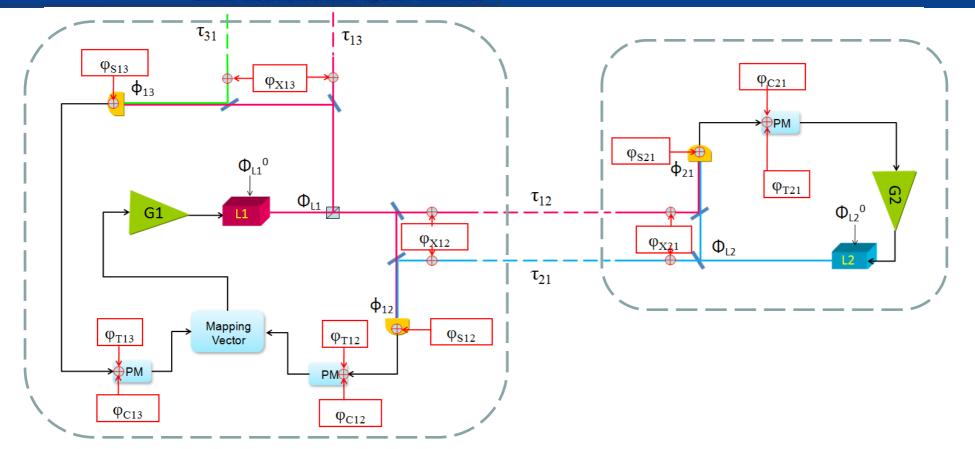


### Noise Model of Arm Locking

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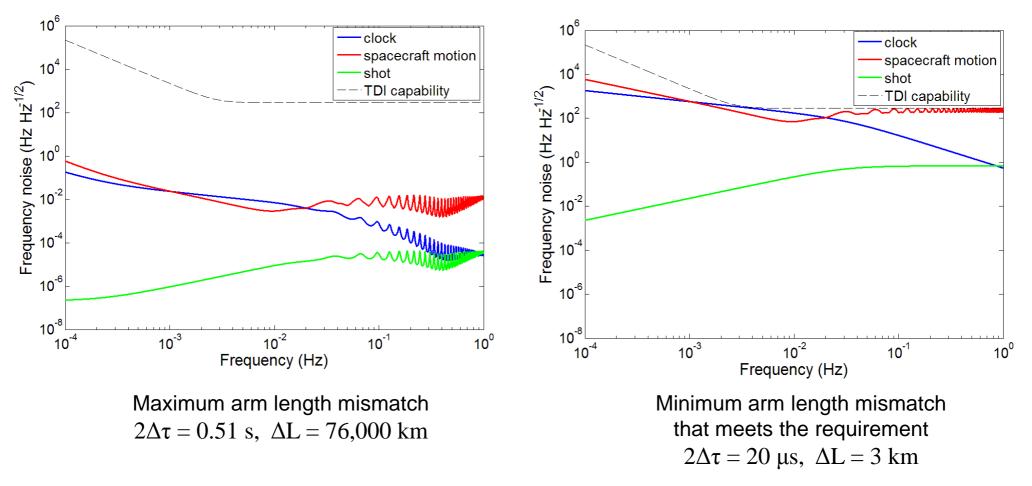
Sclock noise – phase noise of the referencing clock in phase measurements

- SC motion arm length uncertainty from the DRS
- shot noise limited number of photons received per second by photodiodes
- technical noise ADC noise, digitization noise, etc. in signal processing

### Noise Floors of MDAL

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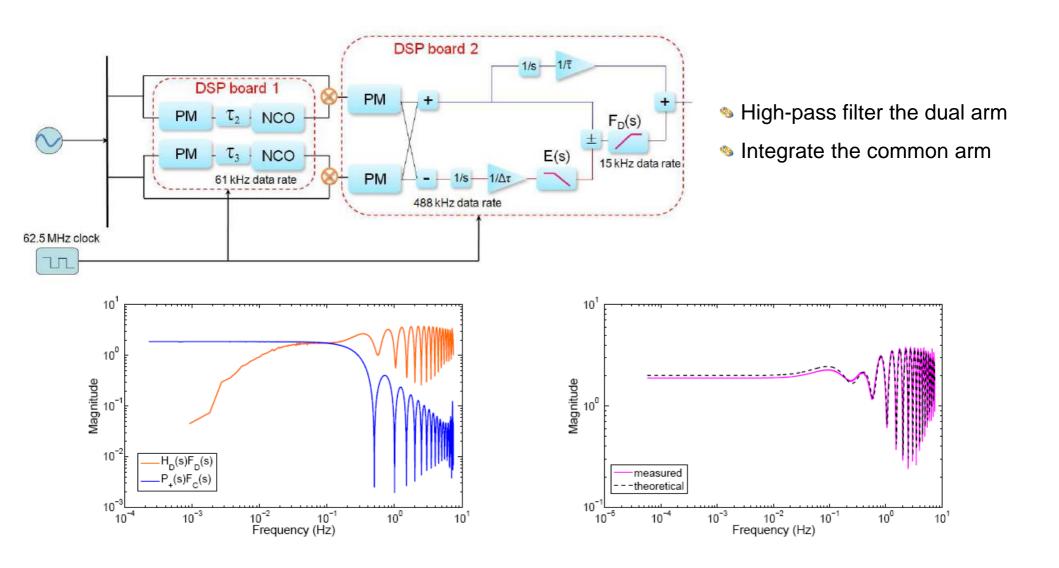
Worse scenario only lasts for ~ 300 s, twice a year.

Arm locking alone can meet the pre-TDI requirement (with all six links operational).



### Control System – Modified Dual

Laser Interferometer Space Antenna



ming a new window on the Universe DIF Laser Interferometer Space Antenna

# MDAL with Pre-stabilized Laser

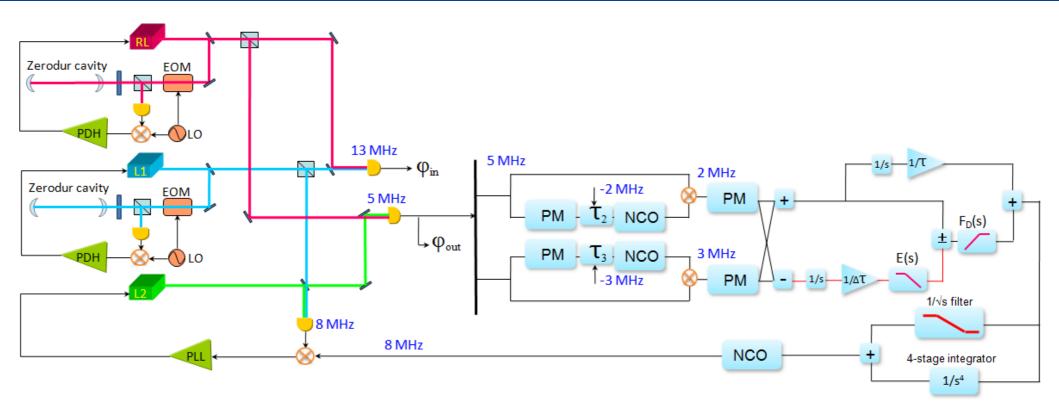
 $V_{\text{SC3}}$ SC3 τ<sub>13</sub> τ<sub>31</sub> EON V<sub>SC2</sub>  $\tau_{12}$  $\square$ 2 τ<sub>21</sub> SC2 sensor SC1 •PM

- Seproduce the noise property via phase locking
- The replica is frequency tunable
- Second states and second st

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# MDAL with Pre-stabilized Laser

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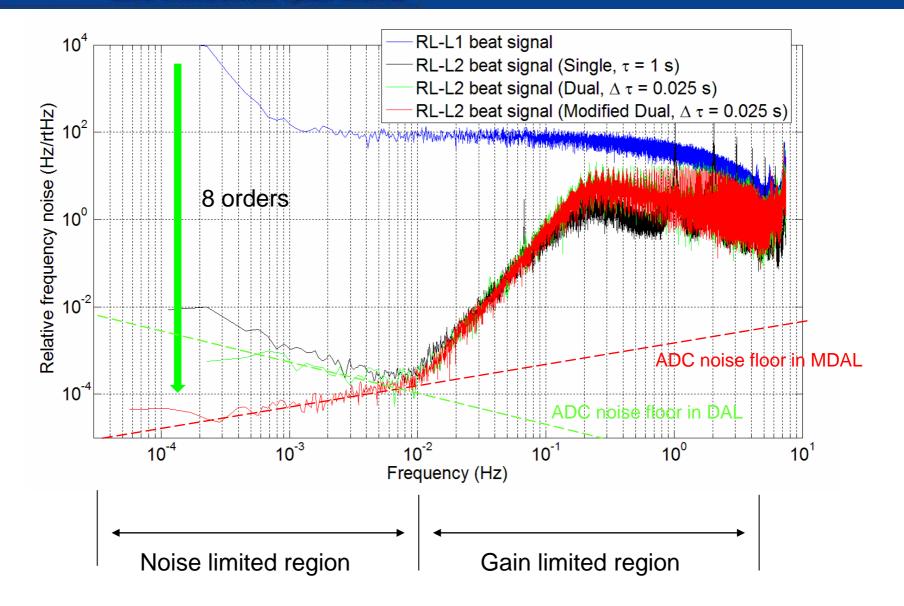
- 48-bit sensor & controller
- $\bar{\tau} = 33 \text{ s}, \Delta \tau = 0.025 \text{ s}$
- Tested with dual & modified dual arm locking

### **Noise Performance**

Laser Interferometer Space Antenna

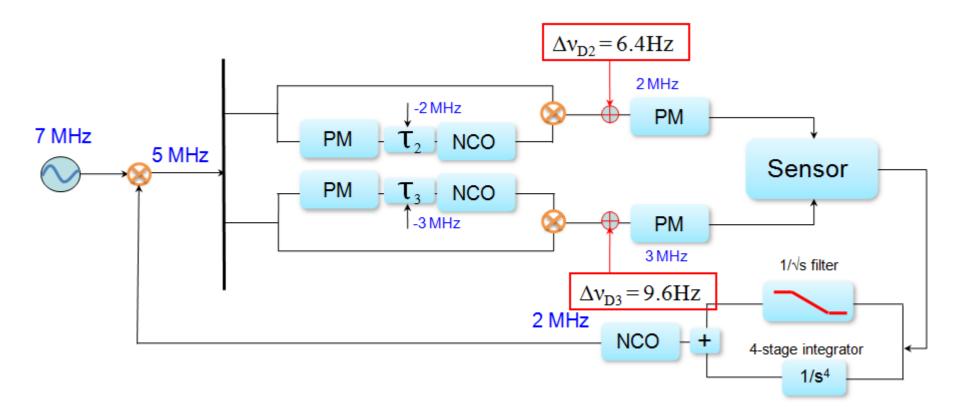
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# Arm Locking with Doppler Errors

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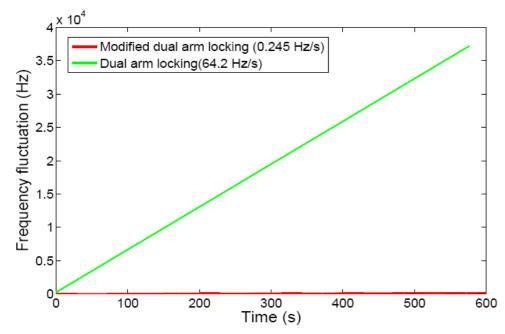
Senerate Doppler frequency error by shifting the clock frequency

$$\bar{\tau} = 33 \text{ s}, \Delta \tau = 0.025 \text{ s}$$

stested with dual & modified dual arm locking

### Drift in Steady State

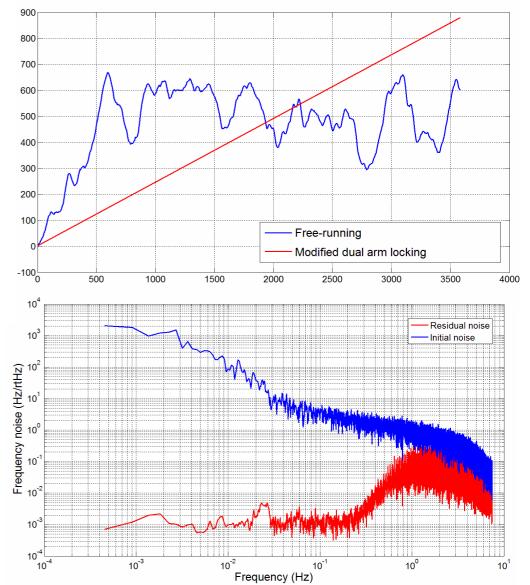
Laser Interferometer Space Antenna



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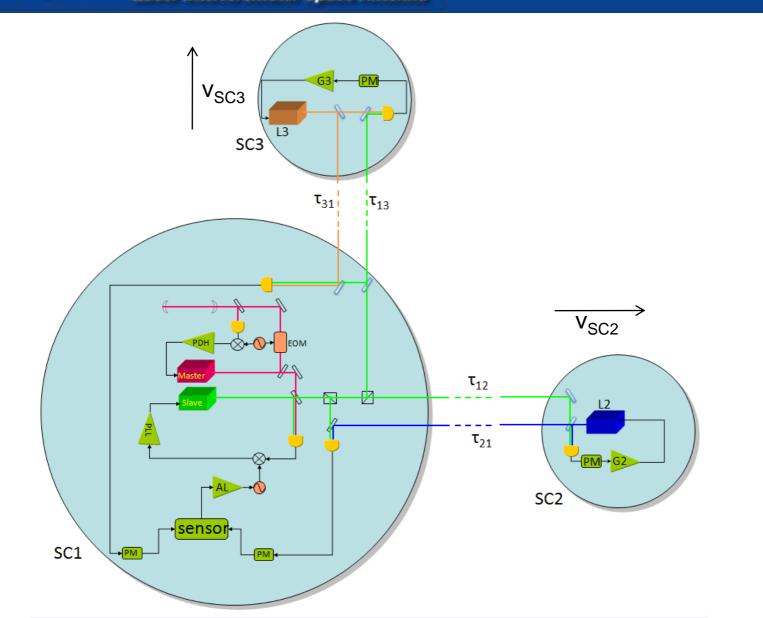
- Sprift rate in dual arm locking:
  - $\frac{\Delta\omega_{\rm D-}}{2\Delta\tau} = 64 \ {\rm Hz/s}$
- Solution State in modified dual arm locking:

$$\frac{\Delta\omega_{\rm D+}}{2\bar{\tau}} = 0.24 \ \rm Hz/s$$



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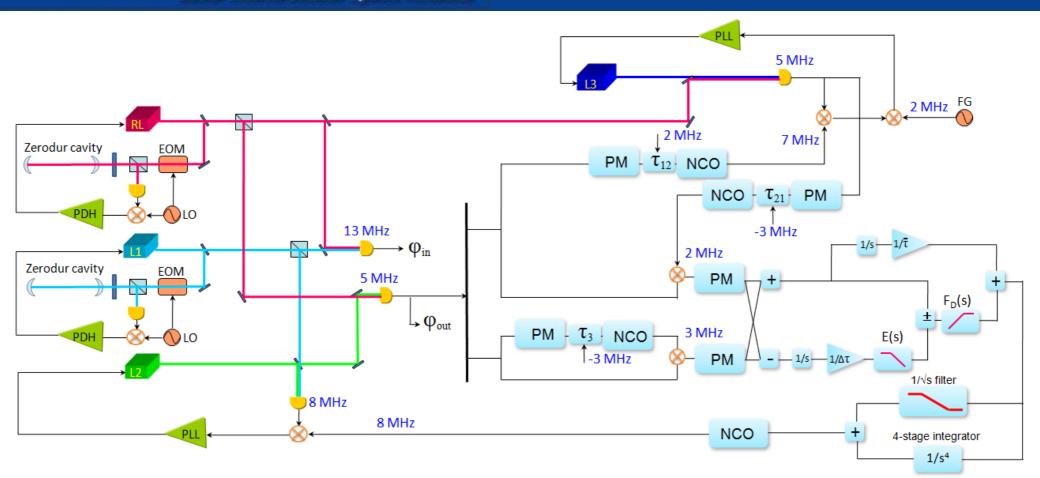
### Put Far-end PLLs Back



### Integrated with Far-end PLL

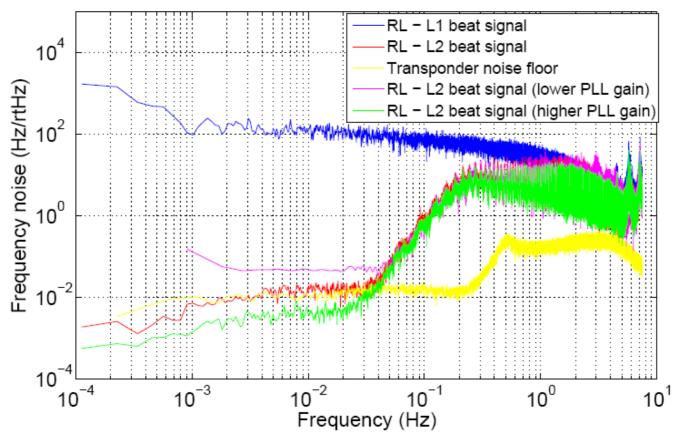
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- Section 2 Phase-lock the far-end laser to the delayed signal
- Sunction generator is synchronized to the master clock
- Transponder noise includes limited PLL gain and clock noise





Results

- Still sufficiently meet the requirement with a margin of 25,000 at 3 mHz.
- The stabilized frequency noise is consistent with the transponder noise floor.
- The noise floor can be adjusted by changing the PLL gain.



# Conclusion

- Sontrol system of arm locking
  - Verified different arm locking schemes under realistic long time delay and Doppler shifts
  - Integrate with cavity stabilized lasers and sufficiently meet the LISA requirement
- Noise limitations of arm locking
  - Dominated by digitization noise or ADC noise in our experiments
  - Demonstrate the noise advantage of modified dual arm locking
  - Demonstrate arm locking performance in the presence of non-negligible transponder noise
- Soppler frequency error in arm locking
  - Studied the frequency pulling in different arm locking schemes

Arm locking has solved the problem of laser frequency stabilization for LISA and can be used in the frequency stabilization for future similar space-based interferometric detectors.



- As announced by ESA in March 2011, due to a modified international cooperation scenario, it is now necessary to study a European-only mission that offers a significant reduction of the cost while maintaining its core science objectives.
- Whilst maximizing the use of results of the LISA studies performed so far and the heritage from LISA Pathfinder, a number of significant changes to the payload, the spacecraft and the mission architecture have to be made to enable the mission to fit the new budget profile.
- The goal is to be ready for a technical and programmatic review starting in November 2011 and for a scientific review by the ESA advisory bodies thereafter, in order to support an SPC downselection for entering phase A/B1 in February 2012.

http://sci.esa.int/science-e/www/object/index.cfm?fobjectid=48728